

2010/ 9/ 22

11th Spacecraft Charging Technology Conference  
New Mexico, USA

Temperature Dependence on Electrostatic  
Discharge in a Simulated Space Plasma  
Environment

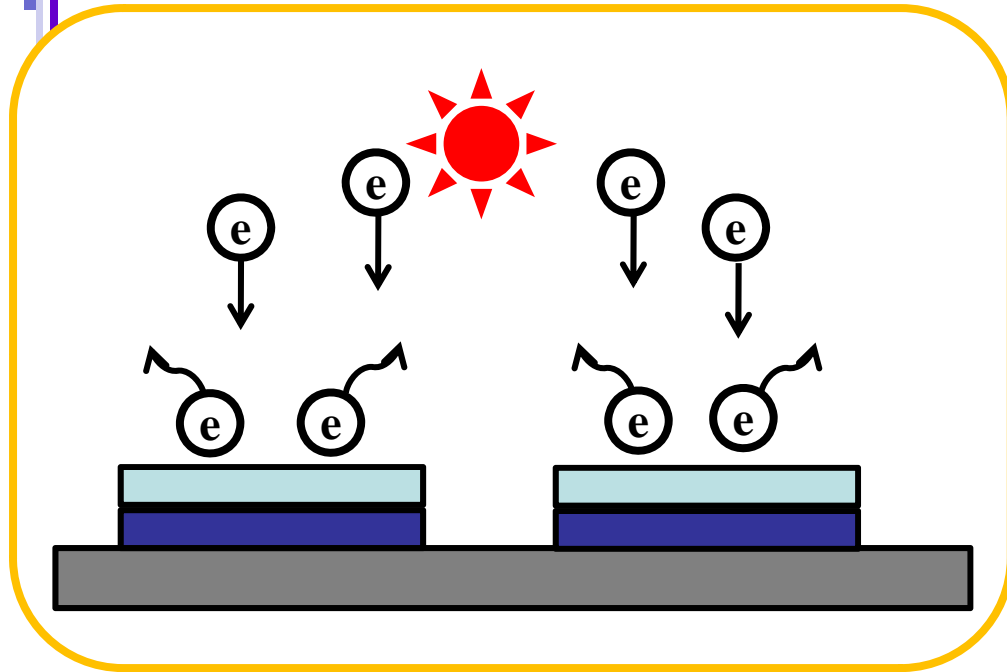
T, Okumura. J, Harada. Y, Hagiwara. K, Nitta. M, Takahashi  
Japan Aerospace Exploration Agency

空へ挑み、宇宙を拓く

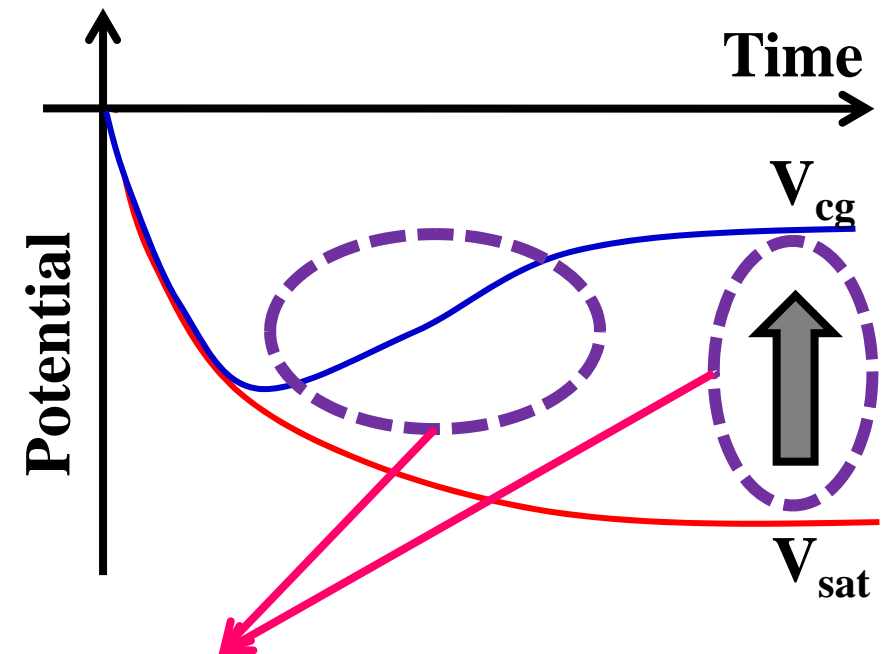


T, Toshimitsu. K, Toyoda  
Kyushu Institute of Technology

No2



## Background

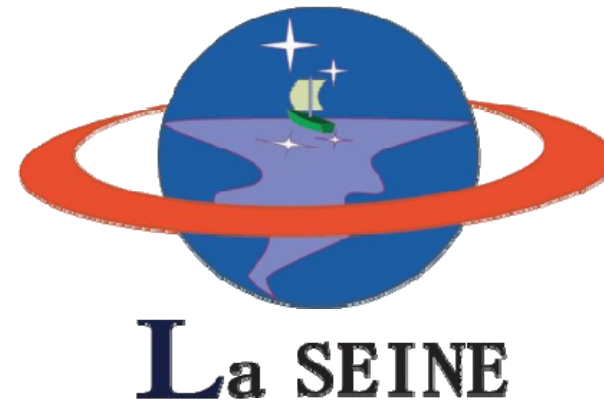


**Depends on temperature??**

- **Discharge number estimation**
- **Proper experiment condition for ESD test**

No3

## Collaboration research



### Temperature dependence on primary discharge

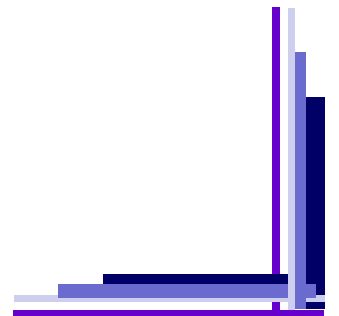
	JAXA	KIT
Threshold voltage	GEO	
Discharge frequency		LEO



*No4*

# **Threshold of differential voltage**

**Experiment technique and result**

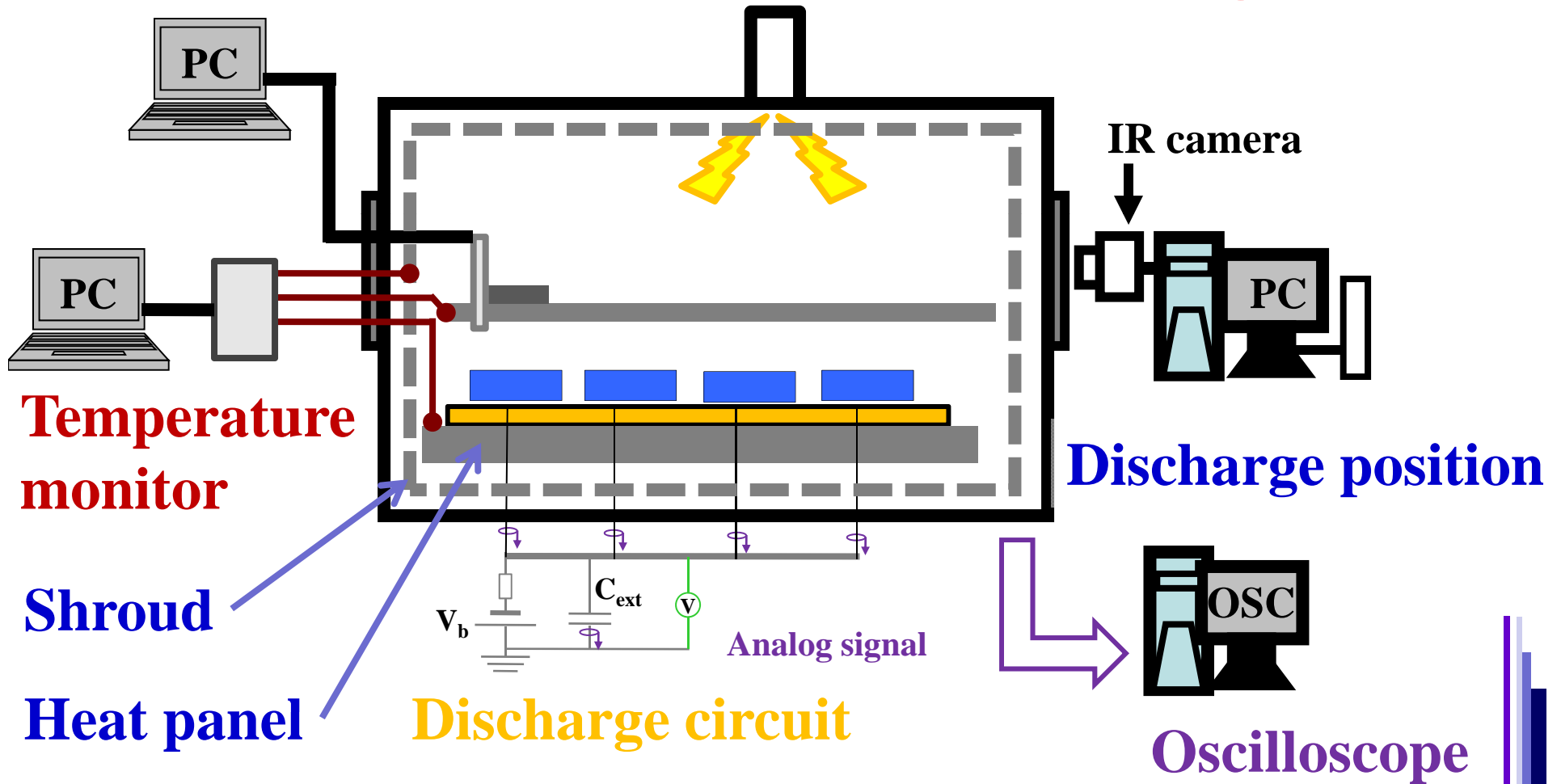


No5

# Experiment system (JAXA)

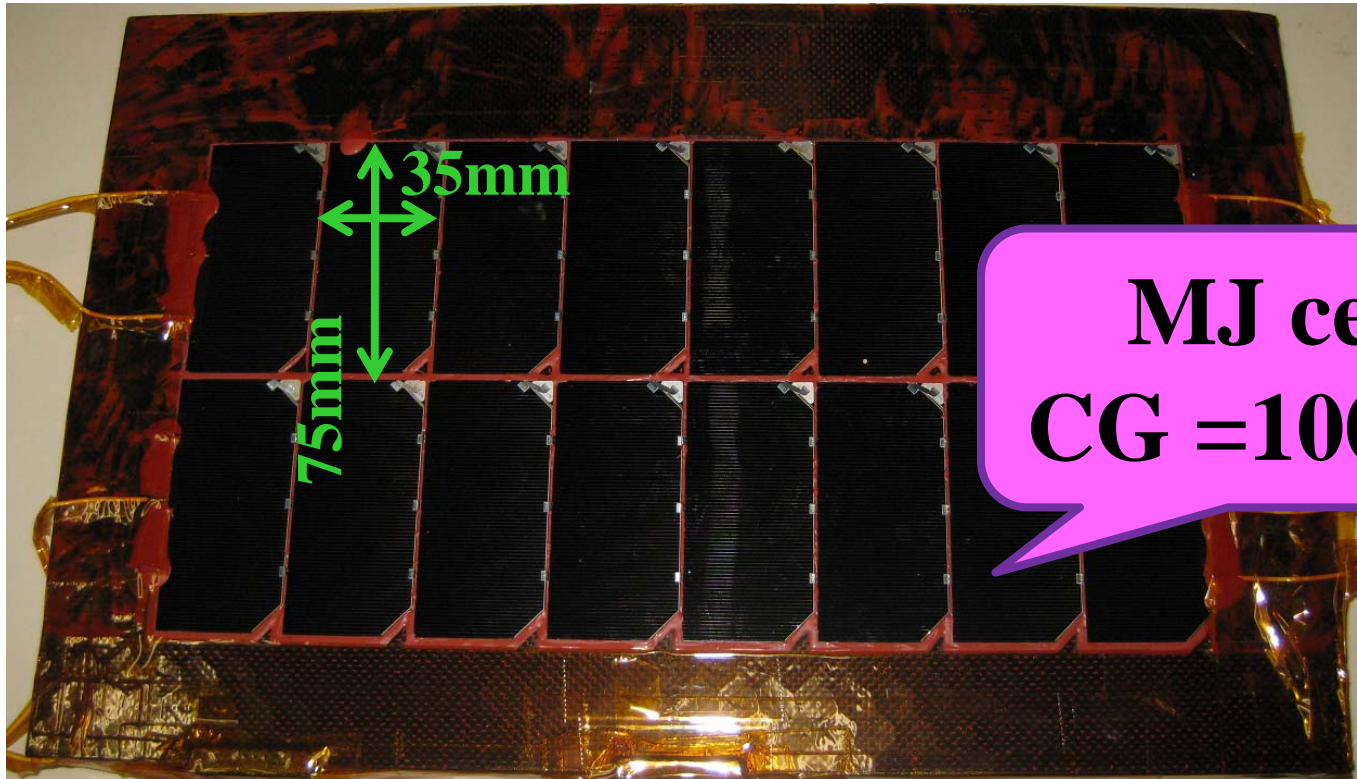
Surface potential meter

Electron beam gun



No6

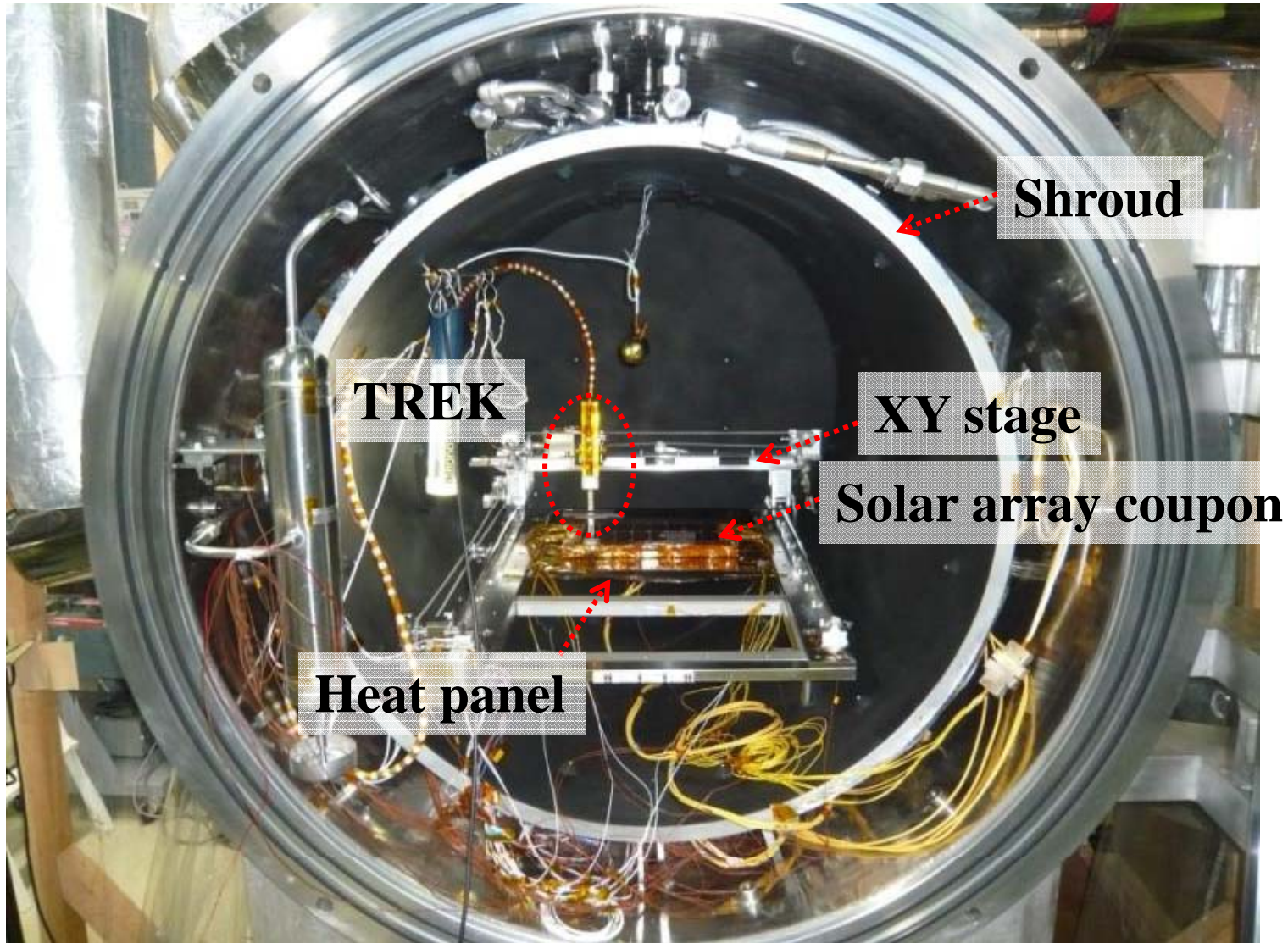
## Solar array coupon



MJ cell  
CG = 100µm

No7

# Experiment system (JAXA)

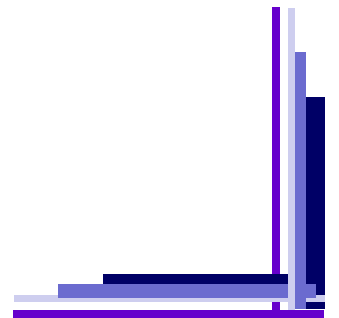




*No8*

## Experiment condition

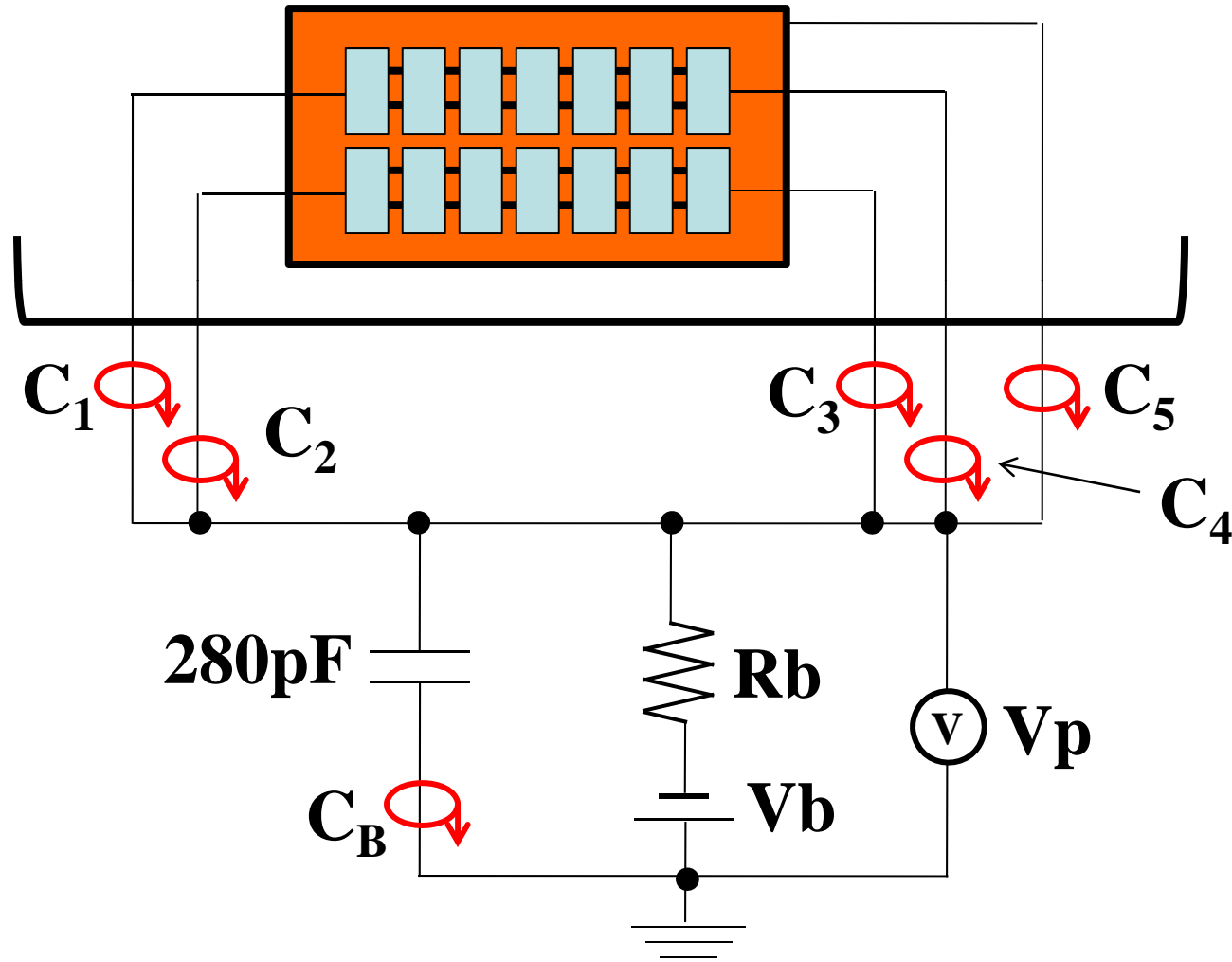
- Pressure:  $5 \times 10^{-5} \text{Pa}$
- Current density:  $4 \text{mA/m}^2$
- Acceleration voltage:  $9 \text{kV}$
- Bias voltage:  $6 \text{kV}$





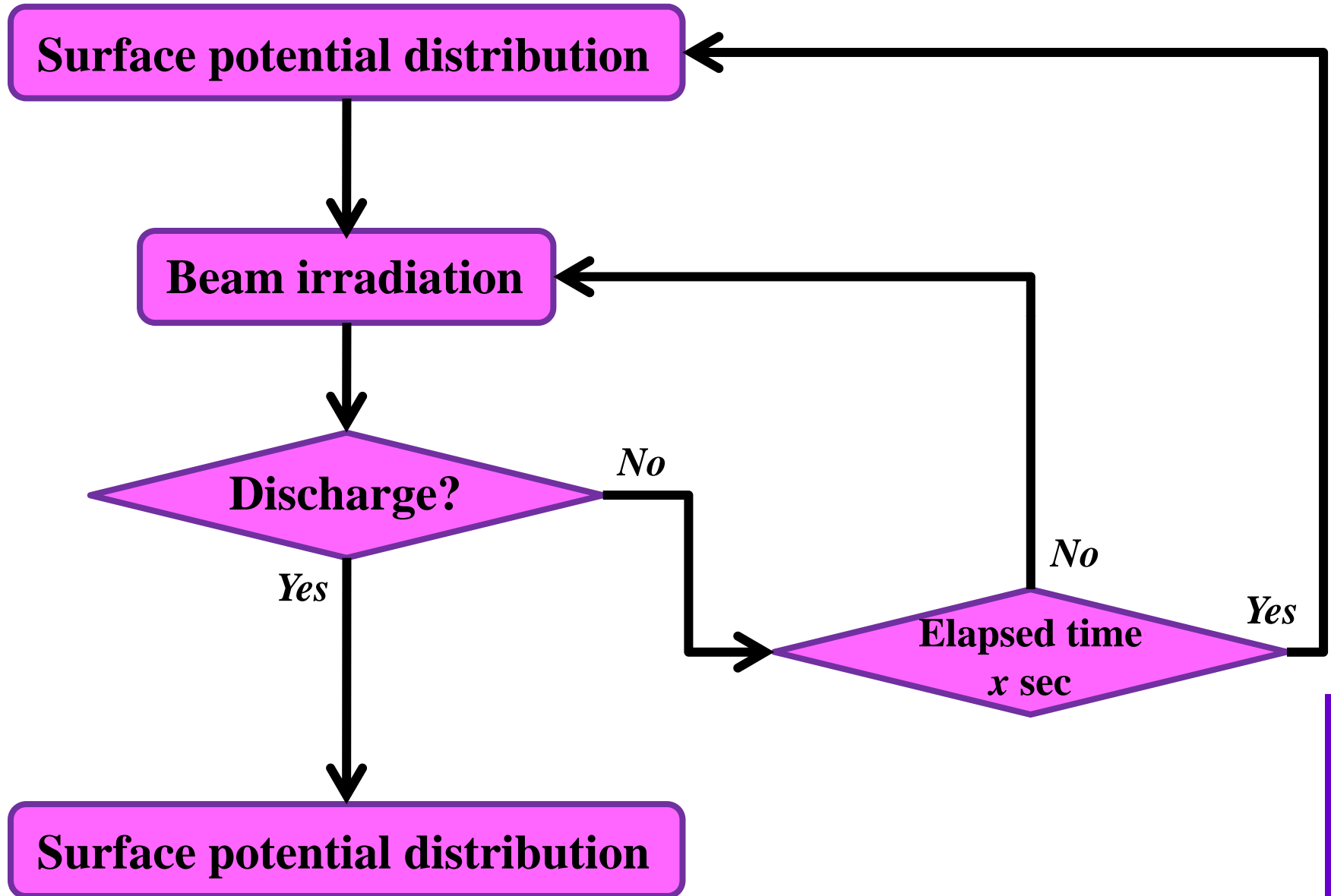
No9

# Discharge circuit



*No10*

# Test sequence

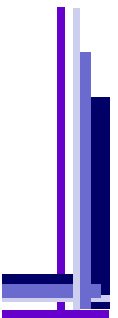
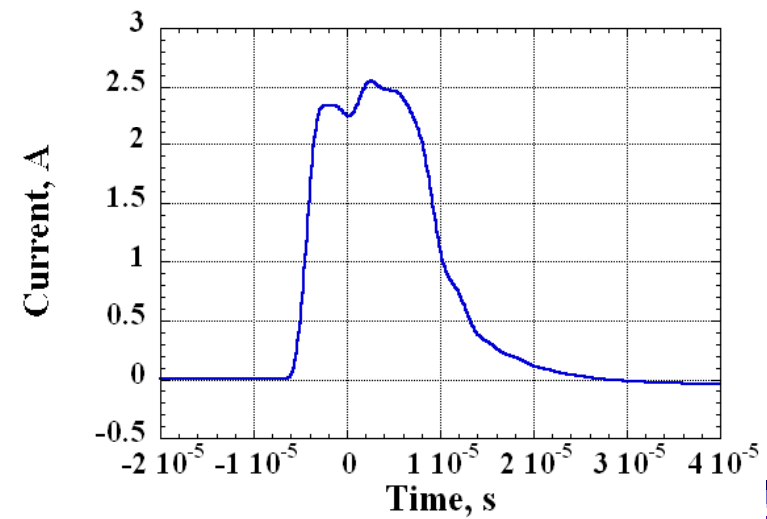
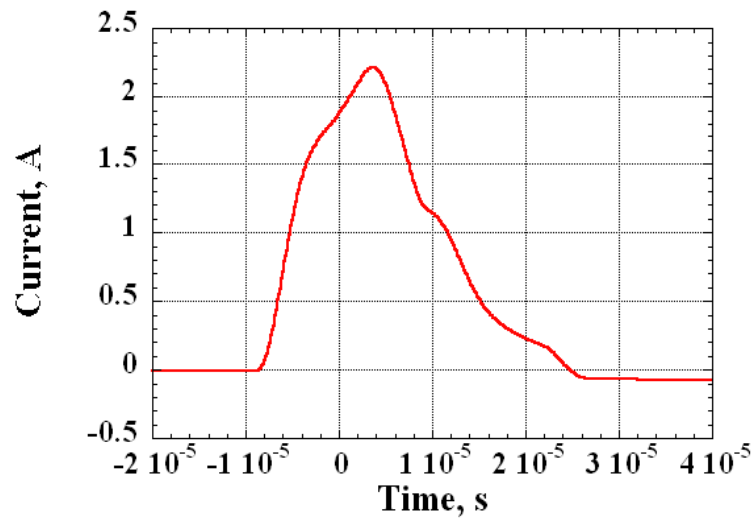
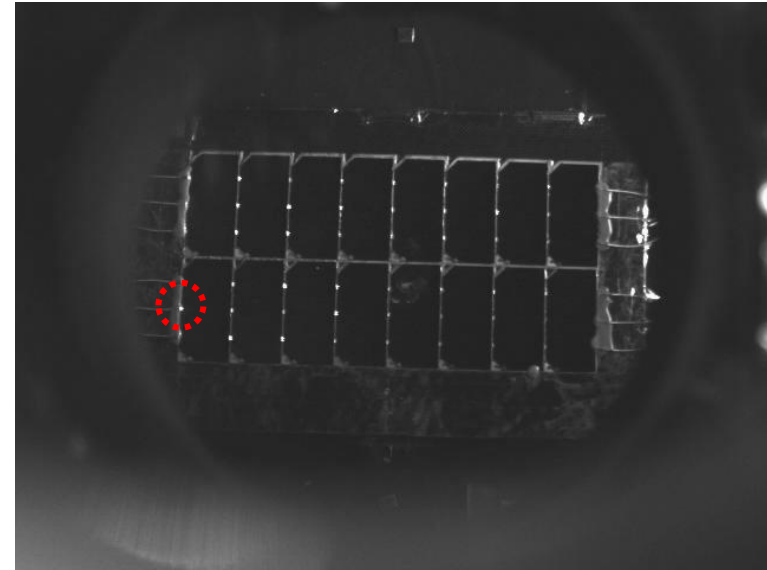
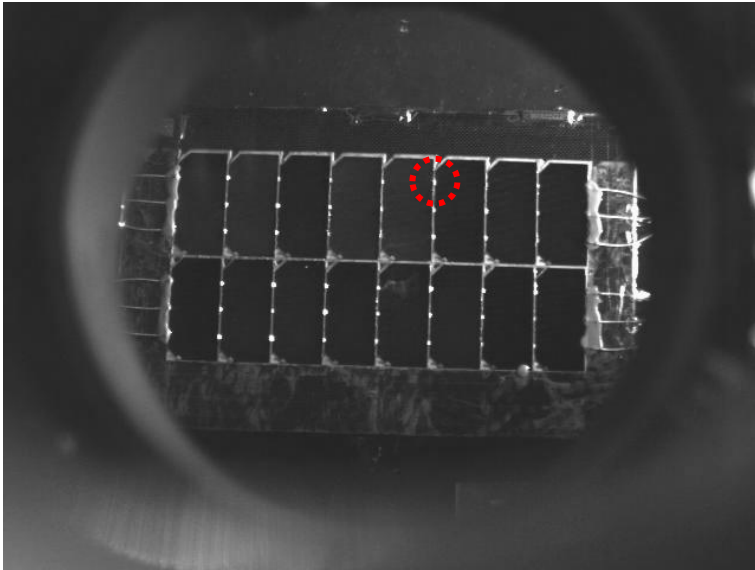


No11

# Discharge position and current waveform

20°C

-110°C

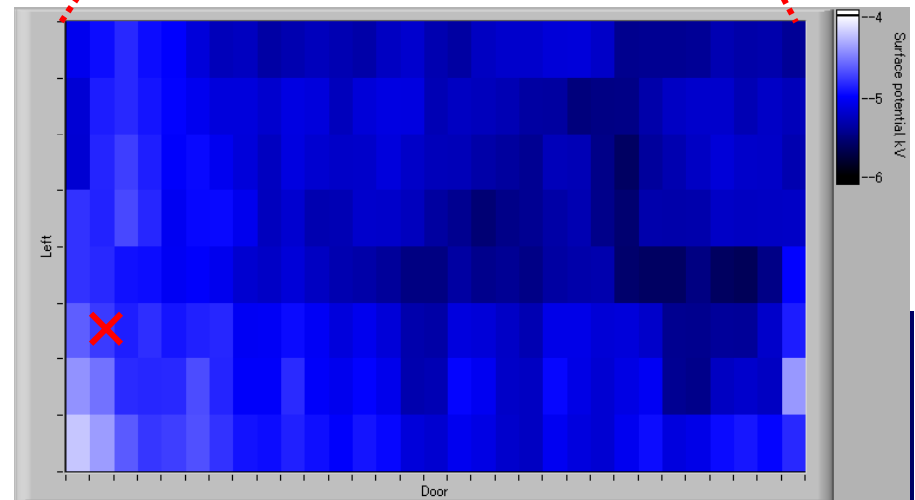
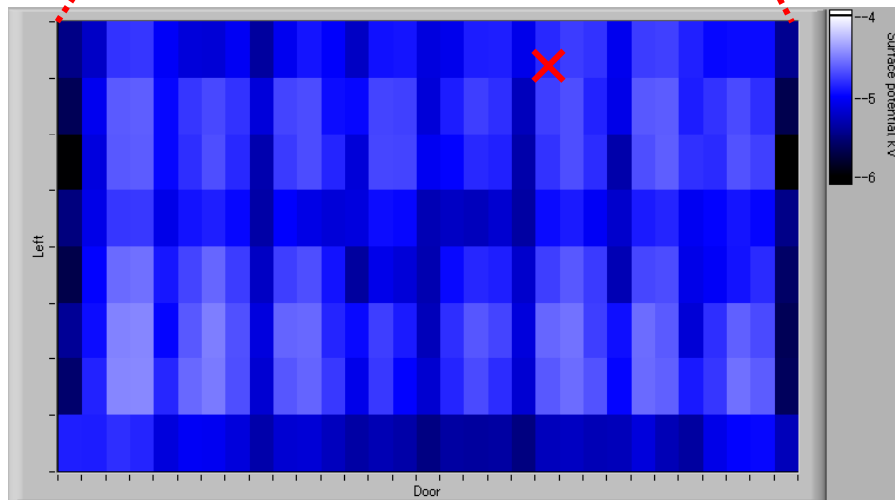
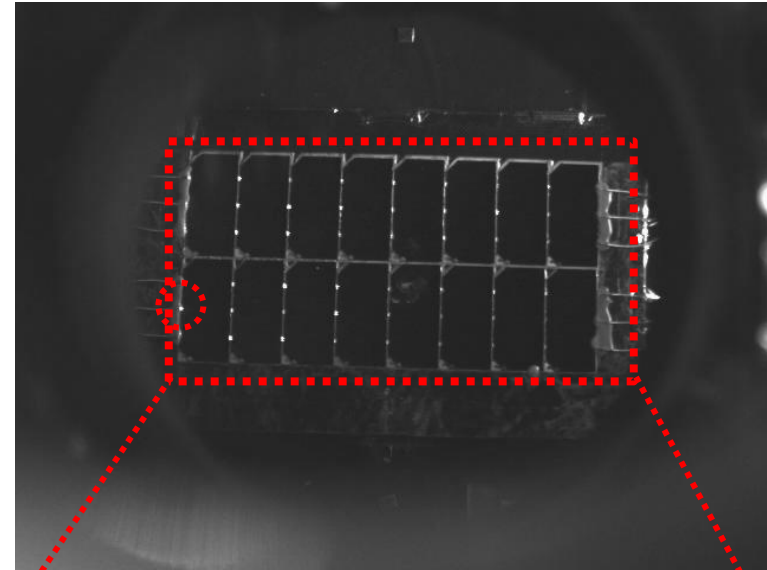
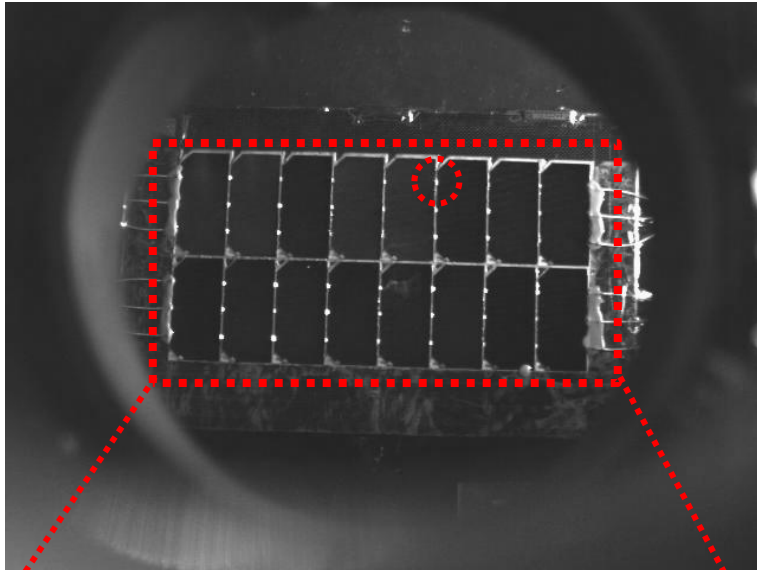


No12

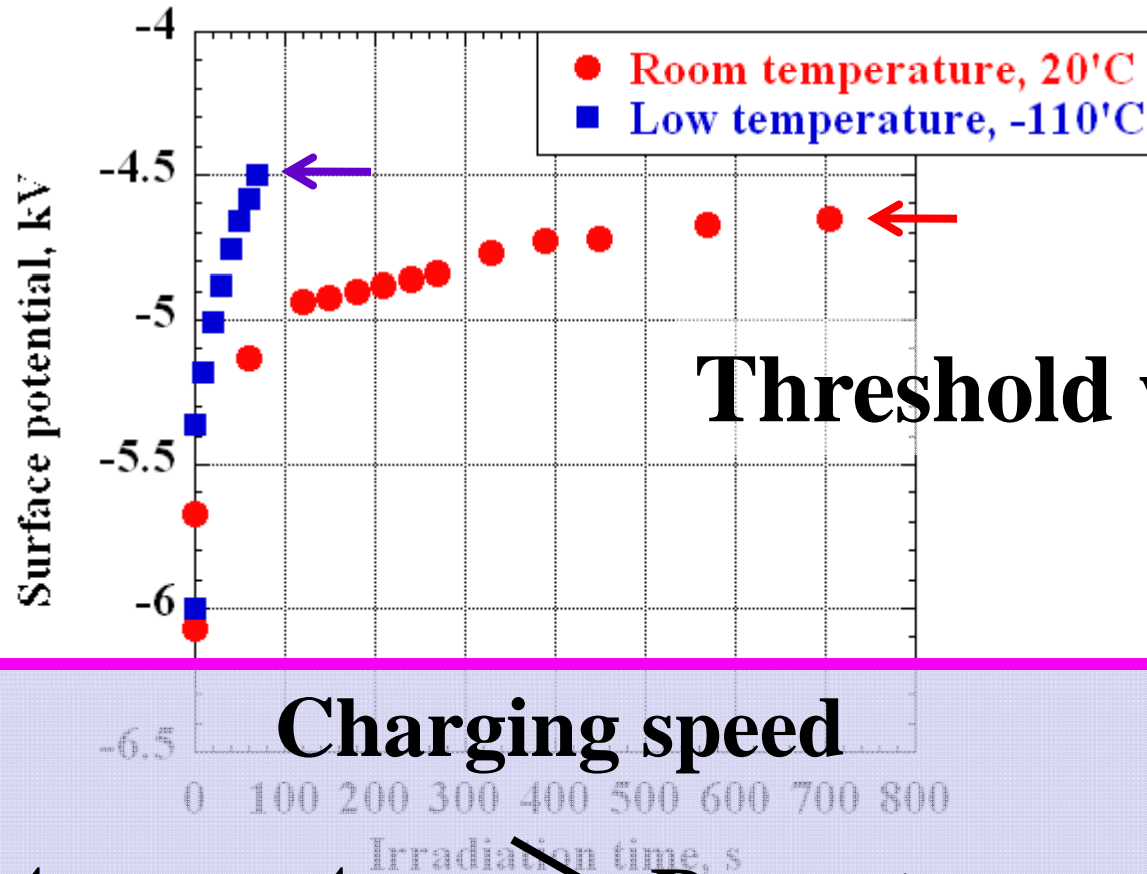
# Discharge position and surface potential

20°C

-110°C



# Surface potential profile



**Charging speed**

**Low temperature > Room temperature**

*No14*

## Threshold volatage

	Room temperature	Low temperature
Threshold voltage	1.3kV	1.4kV
Standard deviation	0.5	0.1

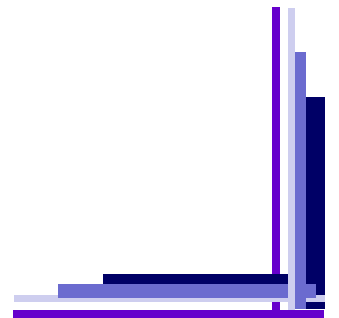
**Threshold voltage does not depend on the temperature**



*No15*

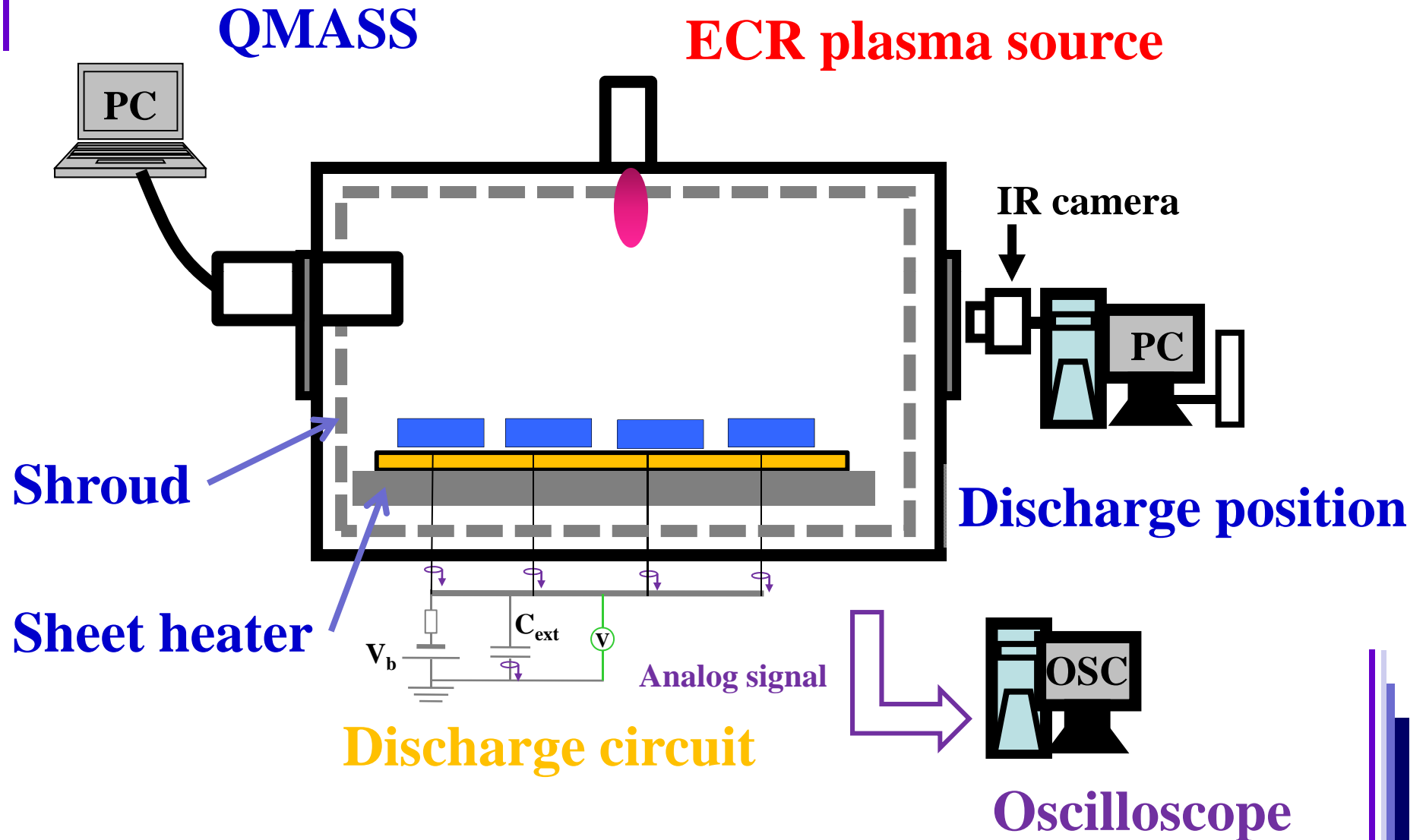
# **Discharge frequency**

**Experiment technique and result**



No16

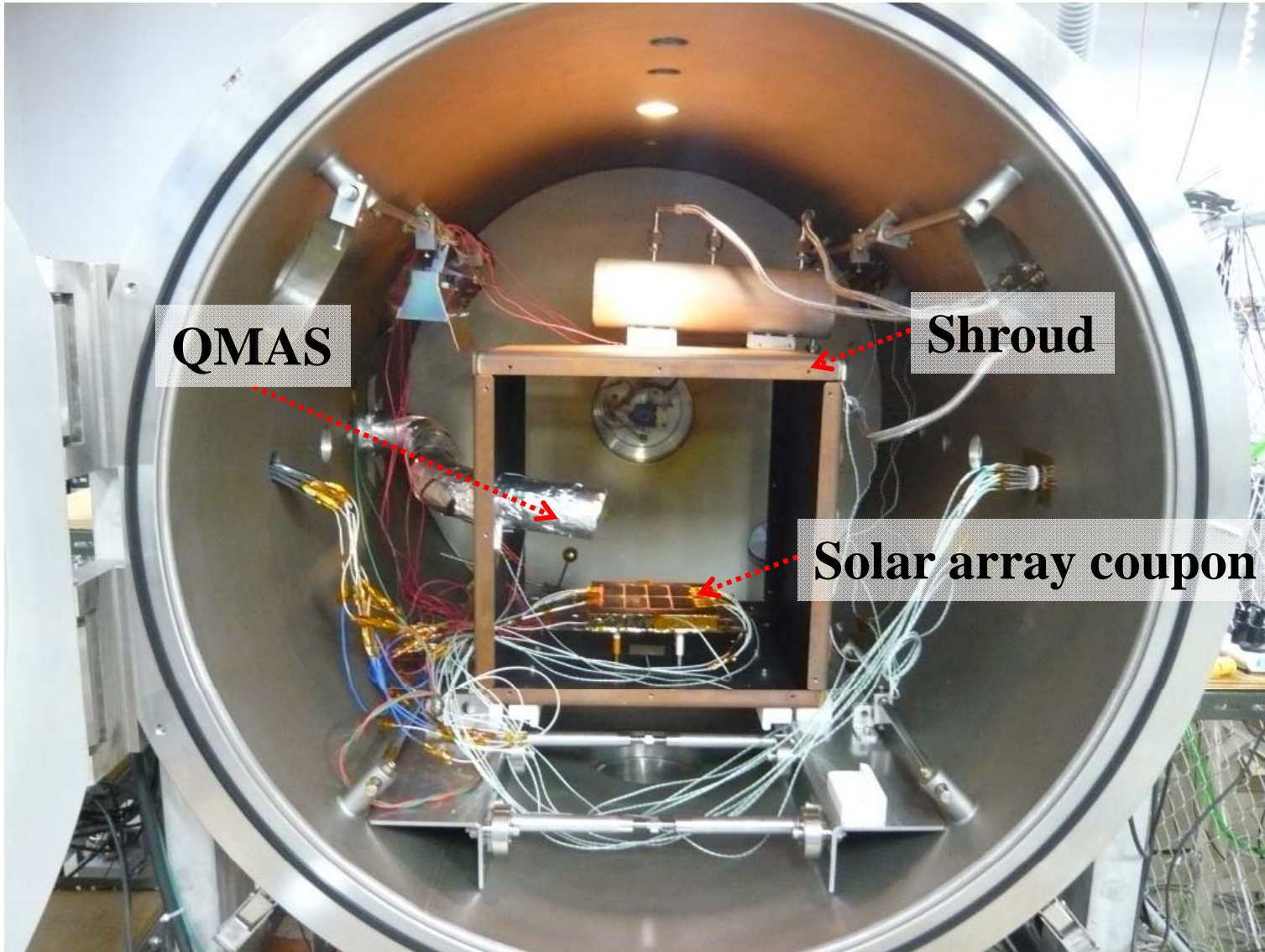
# Experiment system (KIT)





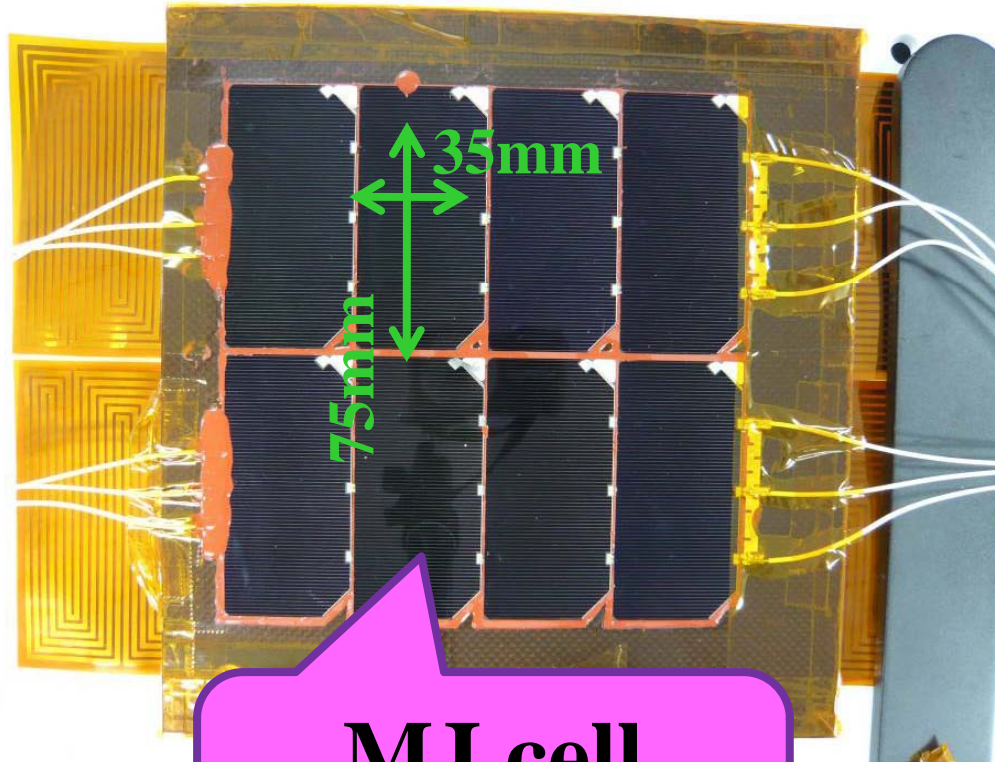
No17

# Experiment system (KIT)



No18

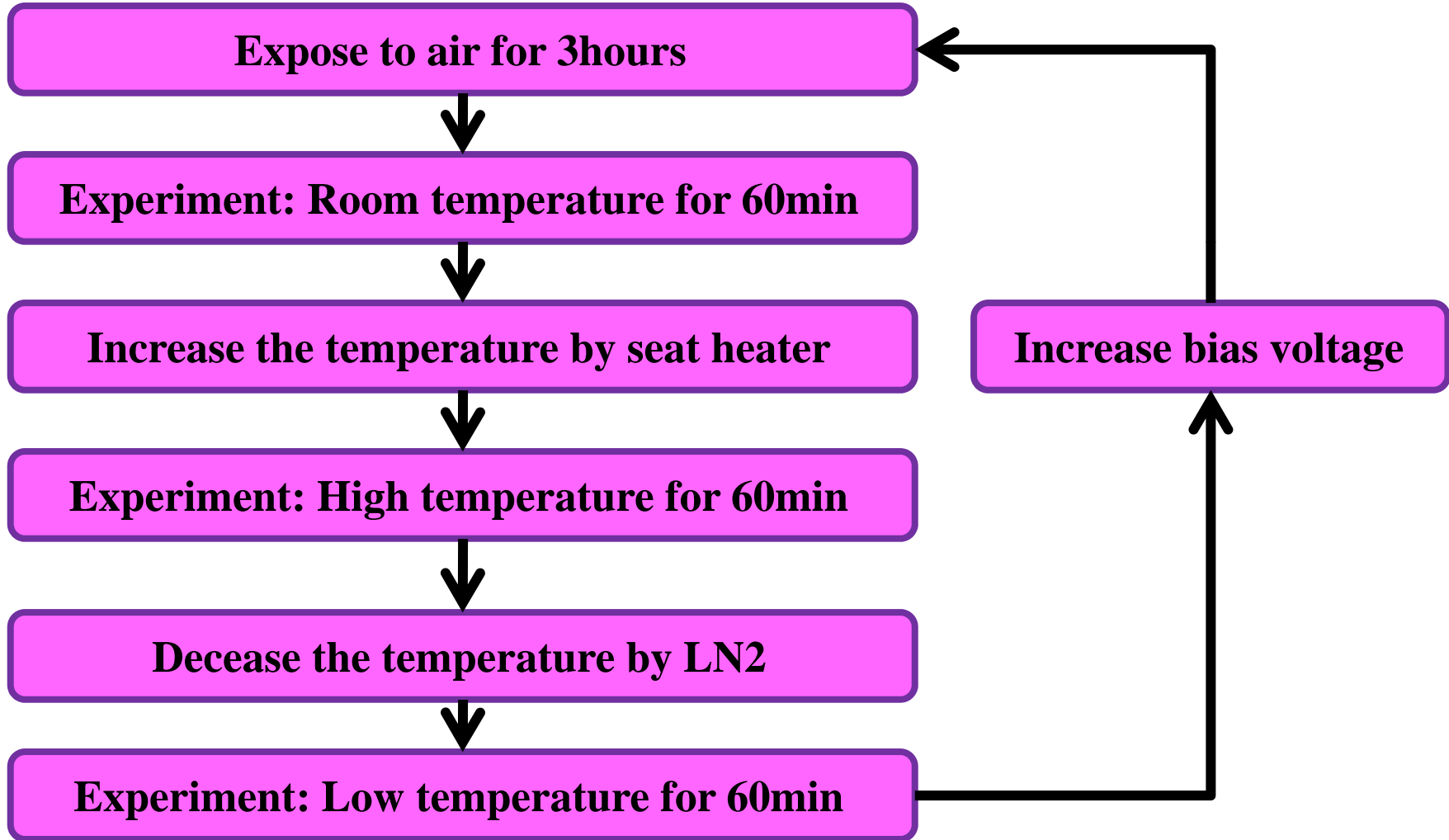
# Solar array coupon



**MJ cell**  
**CG = 100 $\mu$ m**

*No19*

# Test sequence



No20

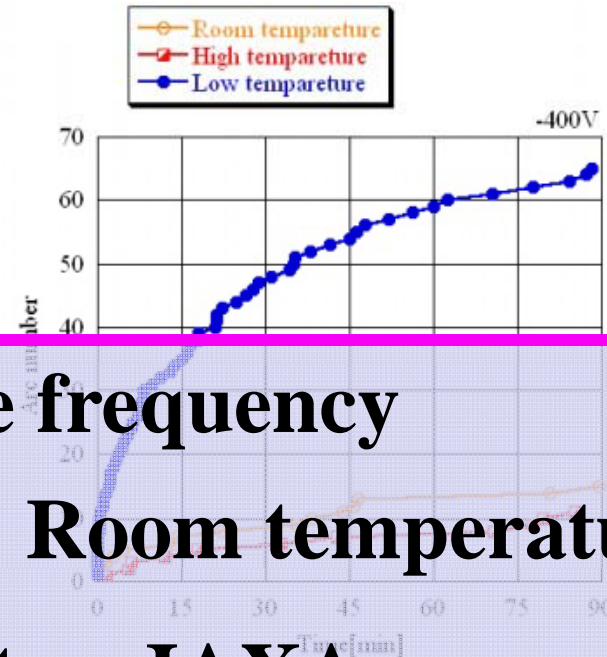
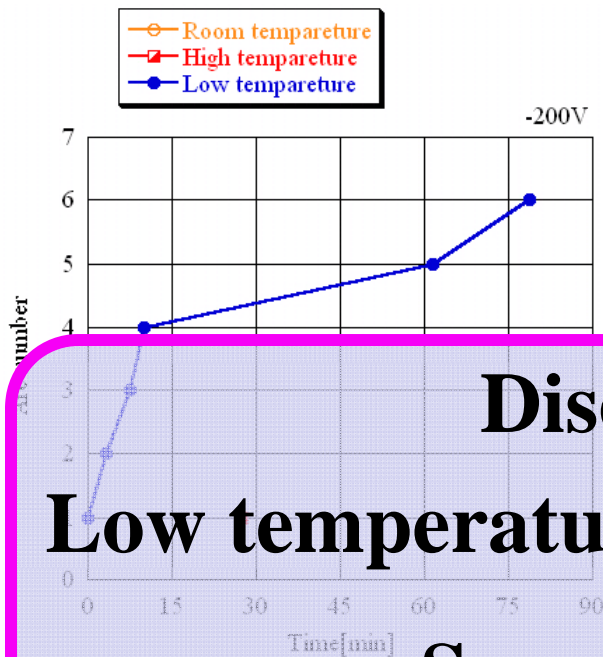
# Experiment condition

$V_{\text{bias}}$ [V]	Experiment time [min]	Ne [1/m <sup>3</sup> ]	Te [eV]	Pressure [Pa]	Coupon temperature [°C]		
					High	room	low
-200	60	$6 \times 10^{12}$	0.7	$2.9 \times 10^{-2}$	64	30	-34
					~	~	~
					66	31	-29
-300	60	$7 \times 10^{12}$	0.7	$2.7 \times 10^{-2}$	64	30	-42
					~	~	~
					65	30	-34
-400	60	$7 \times 10^{12}$	0.7	$3.6 \times 10^{-2}$	63	20	-46
					~	~	~
					65	39	-33

No21

## Temperature dependence on discharge number

Bias	Room(30°C)	High(60°C)	Low(-30°C)
-200	0	1	6
-300	3	9	28
-400	15	11	60

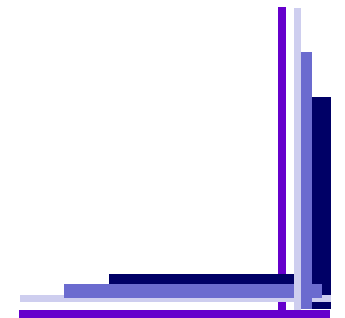


**Discharge frequency**  
**Low temperature > Room temperature**

**Same result as JAXA**

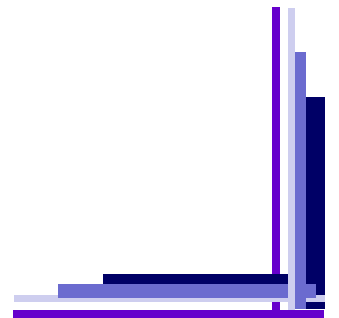
# Summary

- Threshold differential voltage does not depend on temperature in GEO environment
- Arc frequency increases with decreasing temperature in JAXA and KIT



## **Future work**

- ESD test under GEO environment in KIT
- Absorbed material effect on discharge frequency



Thank you for your attention

空へ挑み、宇宙を拓く



宇宙航空研究開発機構  
Japan Aerospace Exploration Agency